

# ON THE DEVELOPMENT OF THE NEW GENERALIZED ORTHOTROPIC DAMAGE AND FRACTURE MODEL eGISSMO

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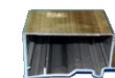
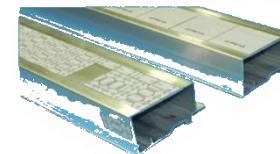
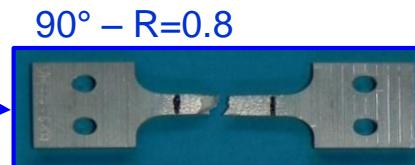
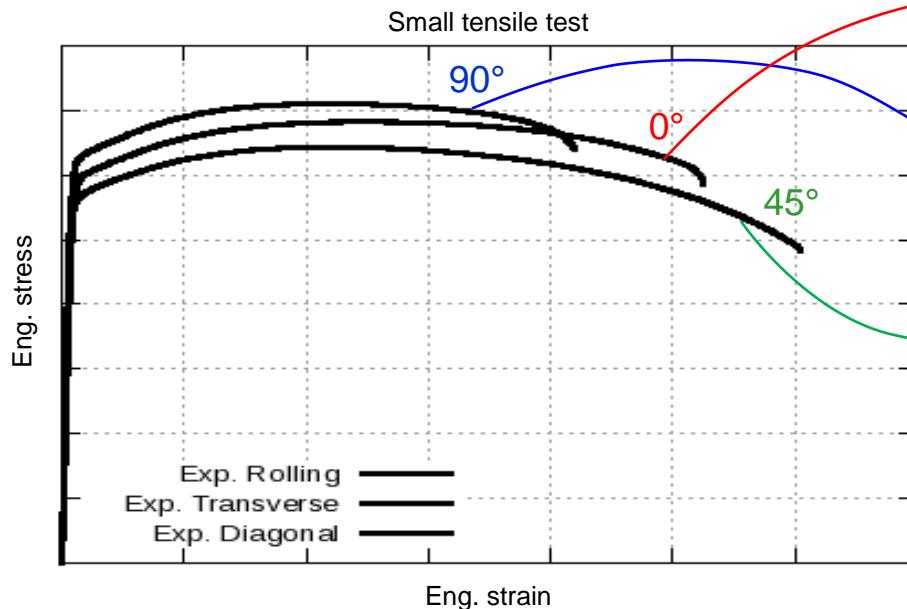
<sup>3</sup>Daimler AG

15<sup>th</sup> German LS-DYNA Forum 2018

Bamberg, October 17, 2018

# Motivation: Orthotropy

Typical aluminum extrusion – mechanical behavior



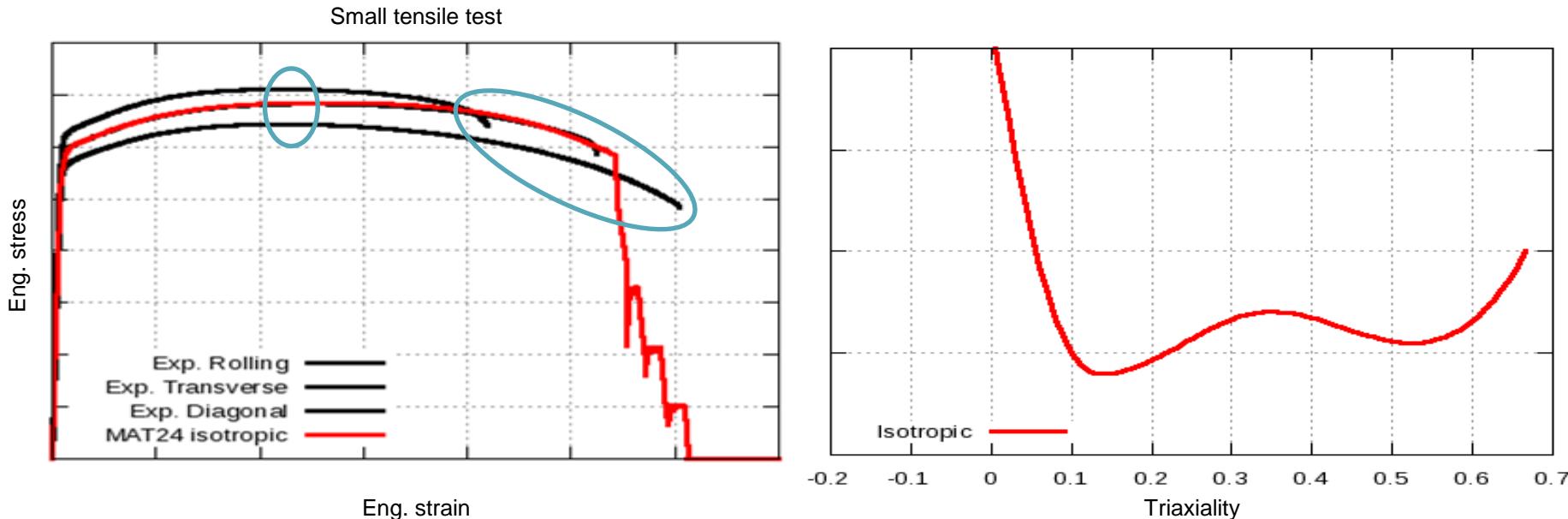
- Stress and failure can be direction-dependent

Fraunhofer  
IWM



# Motivation: Orthotropy

Limitations of isotropic material and failure models (e.g., \*MAT\_024 + GISSMO)



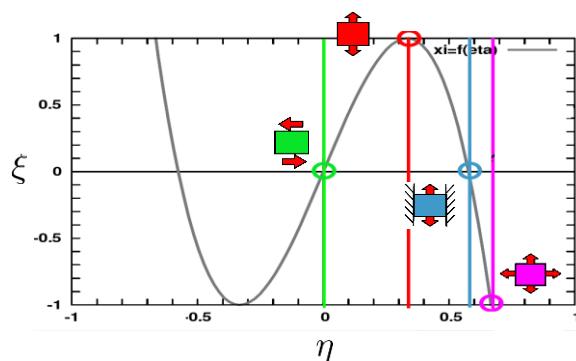
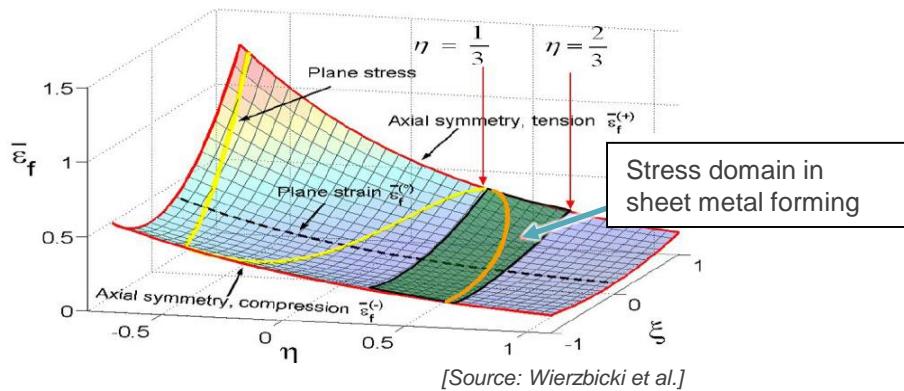
- Different stress levels and fracture strains cannot be captured with isotropic plasticity and isotropic damage/failure models



# **Damage and failure with isotropic GISSMO**

# GISSMO

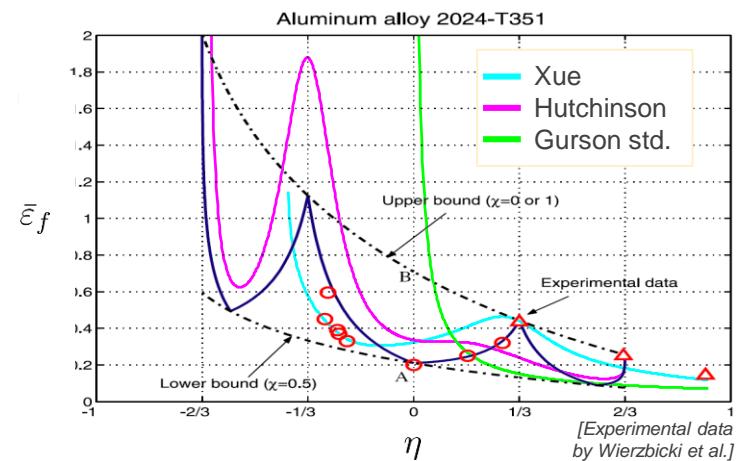
## Failure criterion in planes stress and 3D stress states



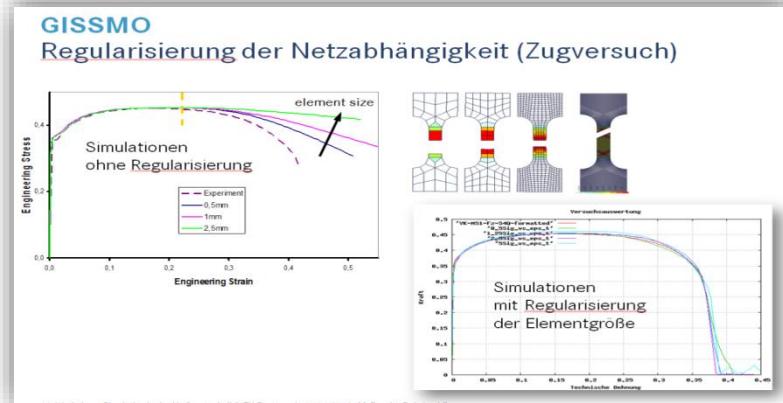
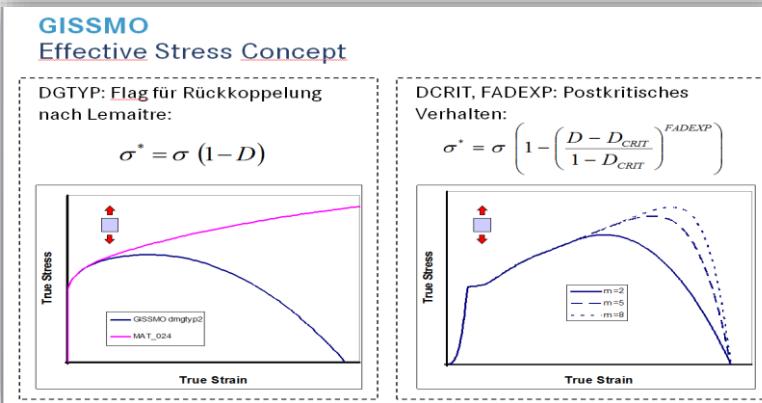
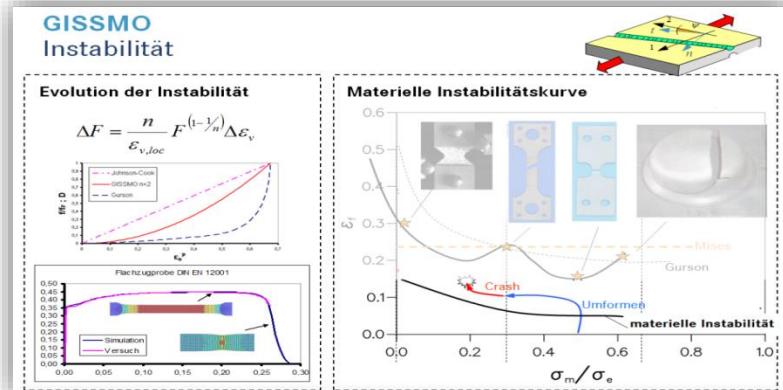
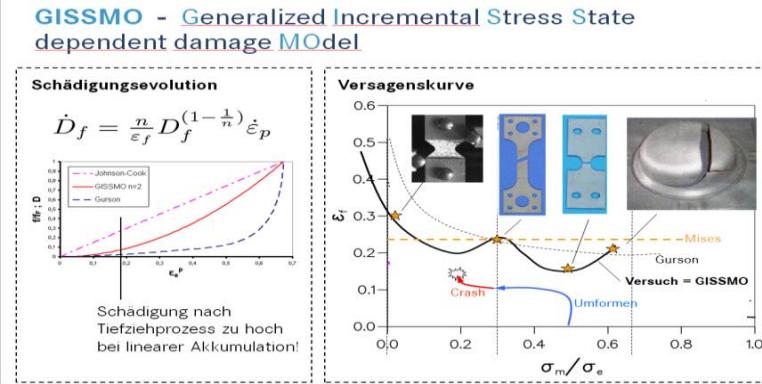
### Parameter definition

$$\eta = \frac{\sigma_m}{\sigma_{vM}} = \frac{I_1}{\sigma_{vM}}$$

$$\xi = \frac{27}{2} \frac{J_3}{(\sigma_{vM})^3} \quad \text{with} \quad J_3 = s_1 s_2 s_3$$



# GISSMO – A quick overview



# Modular Concept

Plasticity model and isotropic damage model: GISSMO

## Plasticity

$$\dot{\sigma}^{eff} = \mathbf{C}(\dot{\varepsilon} - \dot{\varepsilon}_p)$$

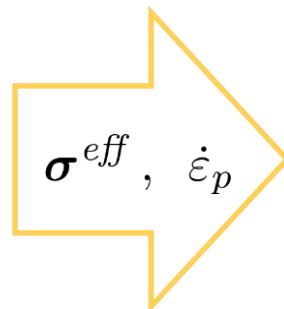
$$\dot{\varepsilon}_p = \lambda \frac{\partial g(\sigma^{eff})}{\partial \sigma^{eff}}$$

$$\dot{\mathbf{q}} = \lambda \frac{\partial \mathbf{q}}{\partial \lambda}$$

$$f(\sigma^{eff}, \mathbf{q}) \leq 0, \quad \dot{\lambda} \geq 0$$

$f(\sigma^{eff})$  = von Mises, Hill, ...

$$g(\sigma^{eff}) = f(\sigma^{eff})$$



## Damage

$$\sigma = \left( 1 - \left( \frac{D - D_{crit}}{1 - D_{crit}} \right)^m \right) \sigma^{eff}$$

$$\dot{F} = n F^{(1-\frac{1}{n})} \frac{\dot{\varepsilon}_p}{\varepsilon_{crit}(\eta, \dot{\varepsilon}_p)}$$

$$\dot{D} = n D^{(1-\frac{1}{n})} \frac{\dot{\varepsilon}_p}{\varepsilon_{fail}(\eta, \xi, l_c, \dot{\varepsilon}_p)}$$

$$F = \int \dot{F} dt \leq 1 \quad \Rightarrow \quad D_{crit} := D$$

$$D = \int \dot{D} dt \leq 1$$



# **Damage and failure with eGISSMO**

# Modular Concept: Toolbox

Plasticity model and anisotropic damage model: eGISSMO

plastic strain tensor  
is estimated:

$$\dot{\varepsilon}_p = \frac{\dot{\varepsilon}_{eff}^p}{\dot{\varepsilon}_{eff}} \left( \dot{\varepsilon} - \frac{\dot{\varepsilon}_{vol}}{3} \delta \right)$$

## Plasticity

$$\dot{\sigma}^{eff} = \mathbf{C}(\dot{\varepsilon} - \dot{\varepsilon}_p)$$

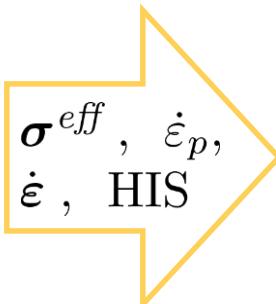
$$\dot{\varepsilon}_p = \dot{\lambda} \frac{\partial g(\sigma^{eff})}{\partial \sigma^{eff}}$$

$$\dot{\mathbf{q}} = \dot{\lambda} \frac{\partial \mathbf{q}}{\partial \lambda}$$

$$f(\sigma^{eff}, \mathbf{q}) \leq 0, \quad \dot{\lambda} \geq 0$$

$f(\sigma^{eff})$  = Barlat, Hill, ...

$$g(\sigma^{eff}) = f(\sigma^{eff})$$



## Damage

$$\sigma = \mathbf{M}^{-1} \sigma^{eff}$$

$$\text{with } \mathbf{M}^{-1} = \begin{bmatrix} \tilde{D}_{11} & \tilde{D}_{12} & \tilde{D}_{13} & 0 & 0 & 0 \\ \tilde{D}_{21} & \tilde{D}_{22} & \tilde{D}_{23} & 0 & 0 & 0 \\ \tilde{D}_{31} & \tilde{D}_{32} & \tilde{D}_{33} & 0 & 0 & 0 \\ 0 & 0 & 0 & \tilde{D}_{44} & 0 & 0 \\ 0 & 0 & 0 & 0 & \tilde{D}_{55} & 0 \\ 0 & 0 & 0 & 0 & 0 & \tilde{D}_{66} \end{bmatrix} \text{ where } \tilde{D}_{ij} = \tilde{D}_{ij}(D_1, D_2, D_3)$$

$$\dot{D}_1 = n_1 D_1^{(1-\frac{1}{n_1})} \frac{\dot{\varepsilon}_1^{eq}}{\varepsilon_1^f(\eta, \xi)}$$

$$\dot{D}_2 = n_2 D_2^{(1-\frac{1}{n_2})} \frac{\dot{\varepsilon}_2^{eq}}{\varepsilon_2^f(\eta, \xi)}$$

$$\dot{D}_3 = n_3 D_3^{(1-\frac{1}{n_3})} \frac{\dot{\varepsilon}_3^{eq}}{\varepsilon_3^f(\eta, \xi)}$$

$$\dot{\varepsilon}_1^{eq} = f_1(\dot{\varepsilon}_{xx}^p, \dot{\varepsilon}_{yy}^p, \dot{\varepsilon}_{zz}^p, \dot{\varepsilon}_{xy}^p, \dot{\varepsilon}_{yz}^p, \dot{\varepsilon}_{xz}^p) \quad \text{or} \quad \text{HIS1}$$

$$\dot{\varepsilon}_2^{eq} = f_2(\dot{\varepsilon}_{xx}^p, \dot{\varepsilon}_{yy}^p, \dot{\varepsilon}_{zz}^p, \dot{\varepsilon}_{xy}^p, \dot{\varepsilon}_{yz}^p, \dot{\varepsilon}_{xz}^p) \quad \text{or} \quad \text{HIS2}$$

$$\dot{\varepsilon}_3^{eq} = f_3(\dot{\varepsilon}_{xx}^p, \dot{\varepsilon}_{yy}^p, \dot{\varepsilon}_{zz}^p, \dot{\varepsilon}_{xy}^p, \dot{\varepsilon}_{yz}^p, \dot{\varepsilon}_{xz}^p) \quad \text{or} \quad \text{HIS3}$$

# Setting up coordinate system for plane stress

- Element coordinate system: a-b
- Material direction=rolling/extrusion direction= x-y
- Principal strains are denoted 1-2

$$(\dot{\varepsilon}_{xx}^p, \dot{\varepsilon}_{yy}^p, \dot{\varepsilon}_{xy}^p) \Rightarrow \left( \dot{\varepsilon}_1^p, b = \frac{\dot{\varepsilon}_2^p}{\dot{\varepsilon}_1^p}, \vartheta \right)$$

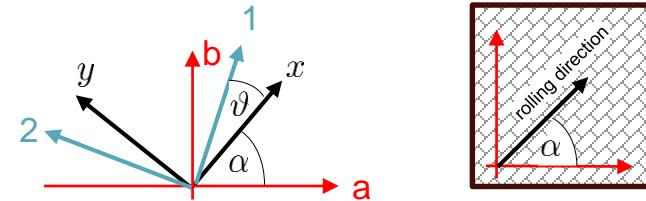
$$\begin{pmatrix} \dot{\varepsilon}_{xx}^p & \dot{\varepsilon}_{xy}^p \\ \dot{\varepsilon}_{xy}^p & \dot{\varepsilon}_{yy}^p \end{pmatrix} = \dot{\varepsilon}_1^p \begin{pmatrix} \cos^2 \vartheta + b \sin^2 \vartheta & (1-b) \sin \vartheta \cos \vartheta \\ (1-b) \sin \vartheta \cos \vartheta & \sin^2 \vartheta + b \cos^2 \vartheta \end{pmatrix}$$

$$\begin{pmatrix} \dot{\varepsilon}_{xx}^p & \dot{\varepsilon}_{xy}^p \\ \dot{\varepsilon}_{xy}^p & \dot{\varepsilon}_{yy}^p \end{pmatrix} = \dot{\varepsilon}_{00}^p \begin{pmatrix} 1 & 0 \\ 0 & b \end{pmatrix} + \dot{\varepsilon}_{90}^p \begin{pmatrix} b & 0 \\ 0 & 1 \end{pmatrix} + \frac{\dot{\varepsilon}_{45}^p}{2} \begin{Bmatrix} \begin{pmatrix} 1+b & 1-b \\ 1-b & 1+b \end{pmatrix} \\ \begin{pmatrix} 1+b & b-1 \\ b-1 & 1+b \end{pmatrix} \end{Bmatrix}$$

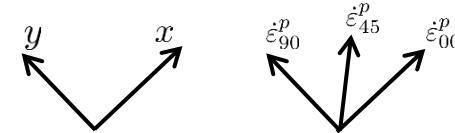
$$\dot{\varepsilon}_{00}^{eq} := 2 |\dot{\varepsilon}_{00}^p| \sqrt{\frac{1}{3}(1+b+b^2)} := 2 |\dot{\varepsilon}_1^p \langle \cos^2 \vartheta - |\cos \vartheta \sin \vartheta| \rangle| \sqrt{\frac{1}{3}(1+b+b^2)}$$

$$\dot{\varepsilon}_{90}^{eq} := 2 |\dot{\varepsilon}_{90}^p| \sqrt{\frac{1}{3}(1+b+b^2)} := 2 |\dot{\varepsilon}_1^p \langle \sin^2 \vartheta - |\cos \vartheta \sin \vartheta| \rangle| \sqrt{\frac{1}{3}(1+b+b^2)}$$

$$\dot{\varepsilon}_{45}^{eq} := 2 |\dot{\varepsilon}_{45}^p| \sqrt{\frac{1}{3}(1+b+b^2)} := 2 |\dot{\varepsilon}_1^p 2 |\cos \vartheta \sin \vartheta|| \sqrt{\frac{1}{3}(1+b+b^2)}$$



Damage accumulation in material coordinate system:



Damage accumulation:

$$D_{00} = \int \dot{D}_{00} dt \quad \dot{D}_{00} = n_{00} D_{00}^{\left(1 - \frac{1}{n_{00}}\right)} \frac{\dot{\varepsilon}_{00}^{eq}}{\dot{\varepsilon}_{00}^f}$$

$$D_{90} = \int \dot{D}_{90} dt \quad \dot{D}_{90} = n_{90} D_{90}^{\left(1 - \frac{1}{n_{90}}\right)} \frac{\dot{\varepsilon}_{90}^{eq}}{\dot{\varepsilon}_{90}^f}$$

$$D_{45} = \int \dot{D}_{45} dt \quad \dot{D}_{45} = n_{45} D_{45}^{\left(1 - \frac{1}{n_{45}}\right)} \frac{\dot{\varepsilon}_{45}^{eq}}{\dot{\varepsilon}_{45}^f}$$

# Orthotropic/isotropic plasticity with orthotropic damage

[DuBois, Erhart, Haufe, Feucht]

plastic strain tensor  
is estimated:

$$\dot{\varepsilon}_p = \frac{\dot{\varepsilon}_{eff}^p}{\dot{\varepsilon}_{eff}} \left( \dot{\varepsilon} - \frac{\dot{\varepsilon}_{vol}}{3} \delta \right)$$

## Plasticity

$$\dot{\sigma}^{eff} = \mathbf{C}(\dot{\varepsilon} - \dot{\varepsilon}_p)$$

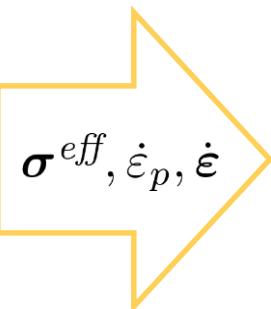
$$\dot{\varepsilon}_p = \dot{\lambda} \frac{\partial g(\sigma^{eff})}{\partial \sigma^{eff}}$$

$$\dot{\mathbf{q}} = \dot{\lambda} \frac{\partial \mathbf{q}}{\partial \lambda}$$

$$f(\sigma^{eff}, \mathbf{q}) \leq 0, \quad \dot{\lambda} \geq 0$$

$$f(\sigma^{eff}) = \text{Barlat, Hill, ...}$$

$$g(\sigma^{eff}) = f(\sigma^{eff})$$



Material coordinate system:

“x – y” → “0° – 90°”

## Damage

$$\sigma = \mathbf{M} \sigma^{eff} \rightarrow \begin{bmatrix} \sigma_{xx} \\ \sigma_{yy} \\ \sigma_{zz=0} \\ \sigma_{xy} \\ \sigma_{yz} \\ \sigma_{xz} \end{bmatrix} = \begin{bmatrix} 1 - \tilde{D}_{00} & 0 & 0 & 0 & 0 & 0 \\ 0 & 1 - \tilde{D}_{90} & 0 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 & 0 & 0 \\ 0 & 0 & 0 & 1 - \tilde{D}_{45} & 0 & 0 \\ 0 & 0 & 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 0 & 0 & 1 \end{bmatrix} \begin{bmatrix} \sigma_{xx}^{eff} \\ \sigma_{yy}^{eff} \\ \sigma_{zz=0}^{eff} \\ \sigma_{xy}^{eff} \\ \sigma_{yz}^{eff} \\ \sigma_{xz}^{eff} \end{bmatrix}$$

$$\dot{D}_{00} = n_{00} D_{00}^{\left(1 - \frac{1}{n_{00}}\right)} \frac{\dot{\varepsilon}_{00}^{eq}}{\varepsilon_{00}^f(\eta)}$$

$$\dot{D}_{90} = n_{90} D_{90}^{\left(1 - \frac{1}{n_{90}}\right)} \frac{\dot{\varepsilon}_{90}^{eq}}{\varepsilon_{90}^f(\eta)}$$

$$\dot{D}_{45} = n_{45} D_{45}^{\left(1 - \frac{1}{n_{45}}\right)} \frac{\dot{\varepsilon}_{45}^{eq}}{\varepsilon_{45}^f(\eta)}$$

$$\text{with } \tilde{D}_i = \left( \frac{D_{(i)} - D_{(i)}^{crit}}{1 - D_{(i)}^{crit}} \right)^{m_i}$$

$$\dot{\varepsilon}_{00}^{eq} = 2 |\dot{\varepsilon}_1^p \langle \cos^2 \vartheta - |\cos \vartheta \sin \vartheta| \rangle| \sqrt{\frac{1}{3}(b^2 + b + 1)}$$

$$\dot{\varepsilon}_{90}^{eq} = 2 |\dot{\varepsilon}_1^p \langle \sin^2 \vartheta - |\cos \vartheta \sin \vartheta| \rangle| \sqrt{\frac{1}{3}(b^2 + b + 1)}$$

$$\dot{\varepsilon}_{45}^{eq} = 2 |\dot{\varepsilon}_1^p 2 |\cos \vartheta \sin \vartheta|| \sqrt{\frac{1}{3}(b^2 + b + 1)}$$

$$\text{either } \max(D_{00}, D_{90}, D_{45}) \leq 1 \quad \text{with} \quad D_i = \int \dot{D}_i dt$$

$$\text{or } D = \int \dot{D} dt \leq 1 \quad \text{with} \quad \dot{D} = \dot{\varepsilon}_p \sqrt{\frac{\dot{D}_{00}^2 + \dot{D}_{90}^2 + \dot{D}_{45}^2}{(\dot{\varepsilon}_{00}^{eq})^2 + (\dot{\varepsilon}_{90}^{eq})^2 + (\dot{\varepsilon}_{45}^{eq})^2}} \quad \left. \right\} \text{according to IFLG3}$$

# Orthotropic damage in plane stress [DuBois et al.]

**IFLG3=0**

## Gissmo 00°

$$\dot{\varepsilon}_{00}^{eq} := 2 \left| \dot{\varepsilon}_1^p \langle \cos^2 \vartheta - |\cos \vartheta \sin \vartheta| \rangle \right| \sqrt{\frac{1}{3}(b^2 + b + 1)}$$

$$\dot{F}_{00} = n_{00} F_{00}^{\left(1 - \frac{1}{n_{00}}\right)} \frac{\dot{\varepsilon}_{00}^{eq}}{\varepsilon_{00}^{crit}(\eta)}; \quad F_{00} = \int \Delta F_{00} \leq 1 \quad \rightarrow \quad D_{00}^{crit} := D_{00}$$

$$\dot{D}_{00} = n_{00} D_{00}^{\left(1 - \frac{1}{n_{00}}\right)} \frac{\dot{\varepsilon}_{00}^{eq}}{\varepsilon_{00}^f(\eta, l_c, \dot{\varepsilon}_{00}^{eq})}; \quad D_{00} = \int \Delta D_{00} \leq 1; \quad \tilde{D}_{00} = \left( \frac{D_{00} - D_{00}^{crit}}{1 - D_{00}^{crit}} \right)^{m_{00}}$$

## Gissmo 90°

$$\dot{\varepsilon}_{90}^{eq} := 2 \left| \dot{\varepsilon}_1^p \langle \sin^2 \vartheta - |\cos \vartheta \sin \vartheta| \rangle \right| \sqrt{\frac{1}{3}(b^2 + b + 1)}$$

$$\dot{F}_{90} = n_{90} F_{90}^{\left(1 - \frac{1}{n_{90}}\right)} \frac{\dot{\varepsilon}_{90}^{eq}}{\varepsilon_{90}^{crit}(\eta)}; \quad F_{90} = \int \Delta F_{90} \leq 1 \quad \rightarrow \quad D_{90}^{crit} := D_{90}$$

$$\dot{D}_{90} = n_{90} D_{90}^{\left(1 - \frac{1}{n_{90}}\right)} \frac{\dot{\varepsilon}_{90}^{eq}}{\varepsilon_{90}^f(\eta, l_c, \dot{\varepsilon}_{90}^{eq})}; \quad D_{90} = \int \Delta D_{90} \leq 1; \quad \tilde{D}_{90} = \left( \frac{D_{90} - D_{90}^{crit}}{1 - D_{90}^{crit}} \right)^{m_{90}}$$

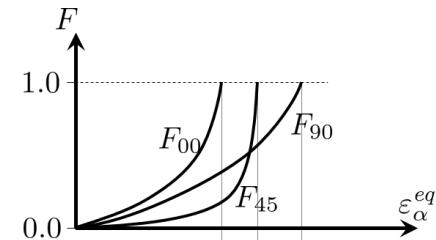
## Gissmo 45°

$$\dot{\varepsilon}_{45}^{eq} := 2 \left| \dot{\varepsilon}_1^p 2 |\cos \vartheta \sin \vartheta| \right| \sqrt{\frac{1}{3}(b^2 + b + 1)}$$

$$\dot{F}_{45} = n_{45} F_{45}^{\left(1 - \frac{1}{n_{45}}\right)} \frac{\dot{\varepsilon}_{45}^{eq}}{\varepsilon_{45}^{crit}(\eta)}; \quad F_{45} = \int \Delta F_{45} \leq 1 \quad \rightarrow \quad D_{45}^{crit} := D_{45}$$

$$\dot{D}_{45} = n_{45} D_{45}^{\left(1 - \frac{1}{n_{45}}\right)} \frac{\dot{\varepsilon}_{45}^{eq}}{\varepsilon_{45}^f(\eta, l_c, \dot{\varepsilon}_{45}^{eq})}; \quad D_{45} = \int \Delta D_{45} \leq 1; \quad \tilde{D}_{45} = \left( \frac{D_{45} - D_{45}^{crit}}{1 - D_{45}^{crit}} \right)^{m_{45}}$$

Example for  $\eta = \text{const.}$



$$\max(D_{00}, D_{90}, D_{45}) \leq 1$$

$$\boldsymbol{\sigma} = \mathbf{M}^{-1} \boldsymbol{\sigma}^{eff}$$

$$\mathbf{M}^{-1} = \begin{bmatrix} 1 - \tilde{D}_{00} & 0 & 0 & 0 & 0 & 0 \\ 0 & 1 - \tilde{D}_{90} & 0 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 & 0 & 0 \\ 0 & 0 & 0 & 1 - \tilde{D}_{45} & 0 & 0 \\ 0 & 0 & 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 0 & 0 & 1 \end{bmatrix}$$

# Orthotropic damage in plane stress [DuBois et al.]

**IFLG3=1**

## Gissmo 00°

$$\dot{\varepsilon}_{00}^{eq} := 2 |\dot{\varepsilon}_1^p \langle \cos^2 \vartheta - |\cos \vartheta \sin \vartheta| \rangle| \sqrt{\frac{1}{3}(b^2 + b + 1)}$$

$$\dot{F}_{00} = n_{00} F_{00}^{\left(1 - \frac{1}{n_{00}}\right)} \frac{\dot{\varepsilon}_{00}^{eq}}{\varepsilon_{00}^{crit}(\eta)}$$

$$\dot{D}_{00} = n_{00} D_{00}^{\left(1 - \frac{1}{n_{00}}\right)} \frac{\dot{\varepsilon}_{00}^{eq}}{\varepsilon_{00}^f(\eta, l_c, \dot{\varepsilon}_{00}^{eq})}; \quad D_{00} = \int \Delta D_{00}; \quad \tilde{D}_{00} = \left( \frac{D - D^{crit}}{1 - D^{crit}} \right)^{m_{00}}$$

## Gissmo 90°

$$\dot{\varepsilon}_{90}^{eq} := 2 |\dot{\varepsilon}_1^p \langle \sin^2 \vartheta - |\cos \vartheta \sin \vartheta| \rangle| \sqrt{\frac{1}{3}(b^2 + b + 1)}$$

$$\dot{F}_{90} = n_{90} F_{90}^{\left(1 - \frac{1}{n_{90}}\right)} \frac{\dot{\varepsilon}_{90}^{eq}}{\varepsilon_{90}^{crit}(\eta)}$$

$$\dot{D}_{90} = n_{90} D_{90}^{\left(1 - \frac{1}{n_{90}}\right)} \frac{\dot{\varepsilon}_{90}^{eq}}{\varepsilon_{90}^f(\eta, l_c, \dot{\varepsilon}_{90}^{eq})}; \quad D_{90} = \int \Delta D_{90}; \quad \tilde{D}_{90} = \left( \frac{D - D^{crit}}{1 - D^{crit}} \right)^{m_{90}}$$

## Gissmo 45°

$$\dot{\varepsilon}_{45}^{eq} := 2 |\dot{\varepsilon}_1^p 2 |\cos \vartheta \sin \vartheta|| \sqrt{\frac{1}{3}(b^2 + b + 1)}$$

$$\dot{F}_{45} = n_{45} F_{45}^{\left(1 - \frac{1}{n_{45}}\right)} \frac{\dot{\varepsilon}_{45}^{eq}}{\varepsilon_{45}^{crit}(\eta)}$$

$$\dot{D}_{45} = n_{45} D_{45}^{\left(1 - \frac{1}{n_{45}}\right)} \frac{\dot{\varepsilon}_{45}^{eq}}{\varepsilon_{45}^f(\eta, l_c, \dot{\varepsilon}_{45}^{eq})}; \quad D_{45} = \int \Delta D_{45}; \quad \tilde{D}_{45} = \left( \frac{D - D^{crit}}{1 - D^{crit}} \right)^{m_{45}}$$

$$\dot{F} = \dot{\varepsilon}_p \sqrt{\frac{\dot{F}_{00}^2 + \dot{F}_{90}^2 + \dot{F}_{45}^2}{(\dot{\varepsilon}_{00}^{eq})^2 + (\dot{\varepsilon}_{90}^{eq})^2 + (\dot{\varepsilon}_{45}^{eq})^2}}$$

$$\rightarrow F = \int \dot{F} dt \leq 1 \rightarrow D^{crit} := D$$

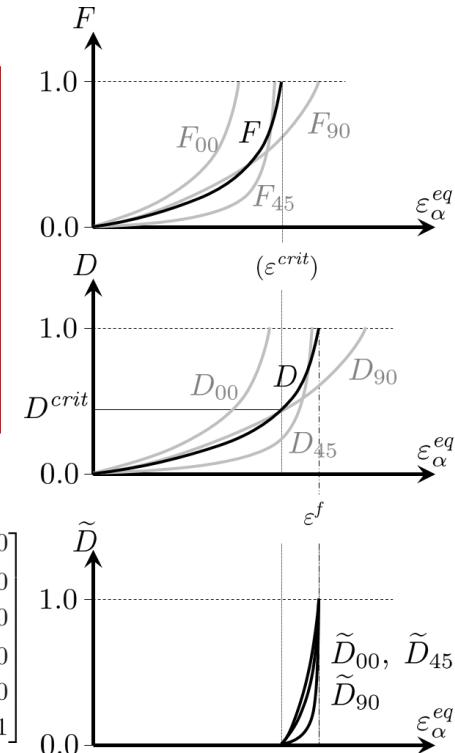
$$\dot{D} = \dot{\varepsilon}_p \sqrt{\frac{\dot{D}_{00}^2 + \dot{D}_{90}^2 + \dot{D}_{45}^2}{(\dot{\varepsilon}_{00}^{eq})^2 + (\dot{\varepsilon}_{90}^{eq})^2 + (\dot{\varepsilon}_{45}^{eq})^2}}$$

$$\rightarrow D = \int \dot{D} dt \leq 1$$

$$\boldsymbol{\sigma} = \mathbf{M}^{-1} \boldsymbol{\sigma}^{eff}$$

$$\mathbf{M}^{-1} = \begin{bmatrix} 1 - \tilde{D}_{00} & 0 & 0 & 0 & 0 & 0 \\ 0 & 1 - \tilde{D}_{90} & 0 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 & 0 & 0 \\ 0 & 0 & 0 & 1 - \tilde{D}_{45} & 0 & 0 \\ 0 & 0 & 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 0 & 0 & 1 \end{bmatrix}$$

Example for  $\eta = \text{const.}$





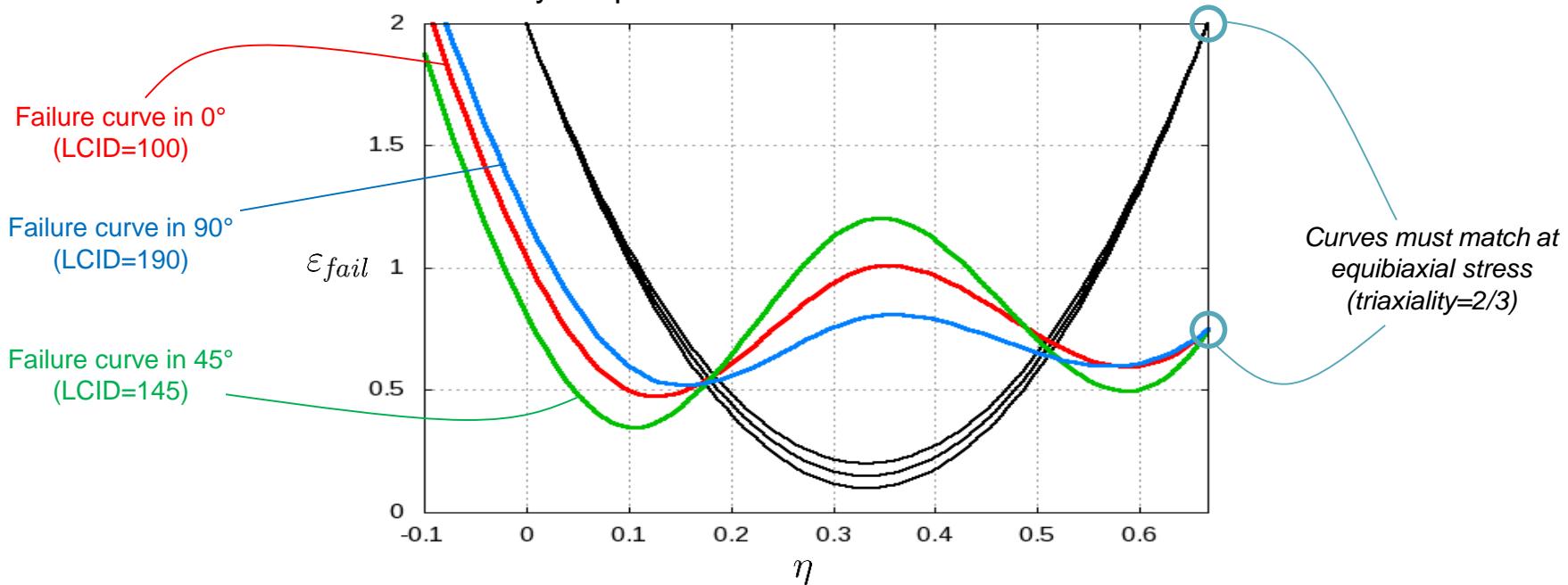
# Application



# eGISSMO in LS-DYNA

Orthotropic damage through **\*MAT\_ADD\_GENERALIZED\_DAMAGE** (eGISSMO)

**Example:** The user can define three different instability (ECRIT) and failure (LCSDG) curves. These curves can have any shape and even cross each other.



# eGISSMO in LS-DYNA

## Orthotropic damage through \*MAT\_ADD\_GENERALIZED\_DAMAGE (eGISSMO)

IDAM=1 → GISSMO is used for  
damage accumulation

DTYP=1 → element  
failure occurs at D=1.0

### \*MAT\_ADD\_GENERALIZED\_DAMAGE

\$	mid	idam	dtyp	refsz	numfip	pddt
\$	10	1	1			
\$	his1	his2	his3	iflg1	iflg2	iflg3
\$	d11	d22	d33	d44	d55	d66
\$	141	142	144	143	144	144
\$	d12	d21	d24	d42	d14	d41

pddt

nhis

Number of history  
variables for the  
accumulation of  
damage

define functions for  
the damage tensor

\$	lcsgd	ecrit	dmgexp	dcrit	fadexp	lcregd
\$	100	-200	2.0		2.5	400
\$	lcsrs	shrf	biaxf			
		1.0	0.0			
\$	lcsgd	ecrit	dmgexp	dcrit	fadexp	lcregd
\$	190	-290	2.0		2.5	490
\$	lcsrs	shrf	biaxf			
		1.0	0.0			
\$	lcsgd	ecrit	dmgexp	dcrit	fadexp	lcregd
\$	145	-245	2.0		2.5	445
\$	lcsrs	shrf	biaxf			
		1.0	0.0			

Rolling/Extrusion  
Direction (0°)

Transverse  
Direction (90°)

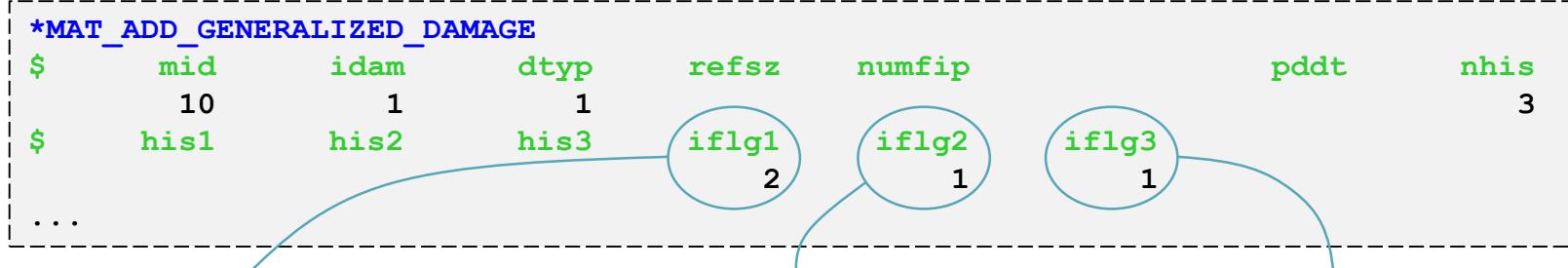
Diagonal  
Direction (45°)

Usual GISSMO  
definitions but now  
in three directions  
(0°, 90°, 45°)

# eGISSMO in LS-DYNA

## Orthotropic damage through \*MAT\_ADD\_GENERALIZED\_DAMAGE (eGISSMO)

\*MAT\_ADD\_GENERALIZED\_DAMAGE is very flexible and has many features embedded.  
For the simulation of orthotropic damage, we currently recommend the following configuration:



### IFLG1=2:

Predefined functions of plastic strain rate components for orthotropic damage.  
IFLG2 should be set to 1.

### IFLG2=1:

The coordinate system for the damage accumulation is the material system. It requires a non-isotropic material model with the AOPT feature

### IFLG3=1:

Erosion occurs when a single damage parameter D reaches unity.

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IFLG1=2, IFLG2=1

```
*MAT_ADD_GENERALIZED_DAMAGE
$      mid      idam      dtyp      refsz      numfip      pddt      nhis
      10        1        1
$      his1      his2      his3      iflg1      iflg2      iflg3
          2        1        1
...
...
```

Predefined functions

0°

$$\Delta\varepsilon_{00}^{eq} = 2 |\Delta\varepsilon_1^p| \langle \cos^2 \vartheta - |\cos \vartheta \sin \vartheta| \rangle \sqrt{\frac{1}{3}(1 + b + b^2)}$$

90°

$$\Delta\varepsilon_{90}^{eq} = 2 |\Delta\varepsilon_1^p| \langle \sin^2 \vartheta - |\cos \vartheta \sin \vartheta| \rangle \sqrt{\frac{1}{3}(1 + b + b^2)}$$

45°

$$\Delta\varepsilon_{45}^{eq} = 4 |\Delta\varepsilon_1^p| |\cos \vartheta \sin \vartheta| \sqrt{\frac{1}{3}(1 + b + b^2)}$$

Instability

$$\Delta F_{00} = n_{00} F_{00}^{\left(1 - \frac{1}{n_{00}}\right)} \frac{\Delta\varepsilon_{00}^{eq}}{\varepsilon_{00}^{crit}(\eta)}$$

$$\Delta F_{90} = n_{90} F_{90}^{\left(1 - \frac{1}{n_{90}}\right)} \frac{\Delta\varepsilon_{90}^{eq}}{\varepsilon_{90}^{crit}(\eta)}$$

$$\Delta F_{45} = n_{45} F_{45}^{\left(1 - \frac{1}{n_{45}}\right)} \frac{\Delta\varepsilon_{45}^{eq}}{\varepsilon_{45}^{crit}(\eta)}$$

Damage

$$\Delta D_{00} = n_{00} D_{00}^{\left(1 - \frac{1}{n_{00}}\right)} \frac{\Delta\varepsilon_{00}^{eq}}{\varepsilon_{00}^f(\eta)}$$

$$\Delta D_{90} = n_{90} D_{90}^{\left(1 - \frac{1}{n_{90}}\right)} \frac{\Delta\varepsilon_{90}^{eq}}{\varepsilon_{90}^f(\eta)}$$

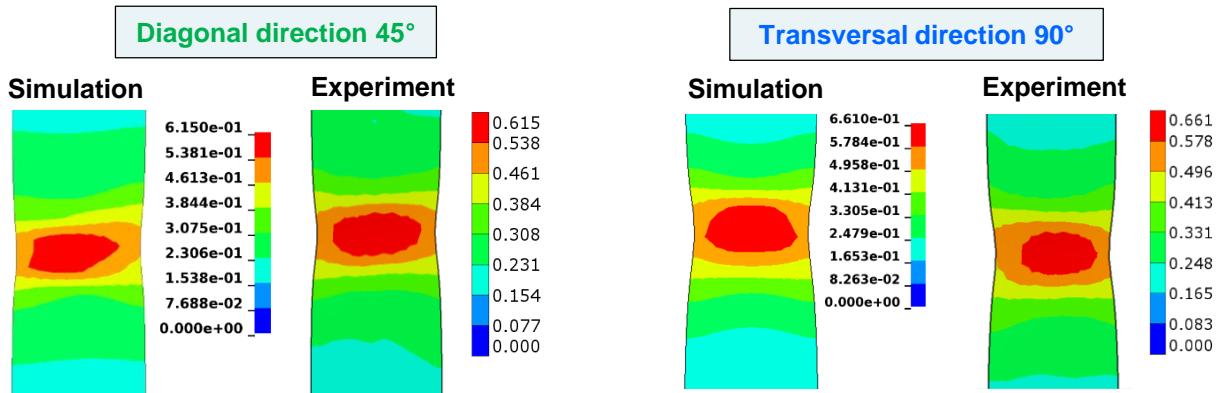
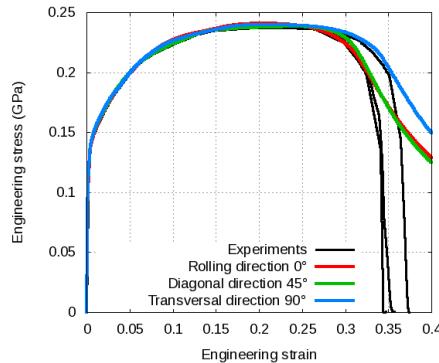
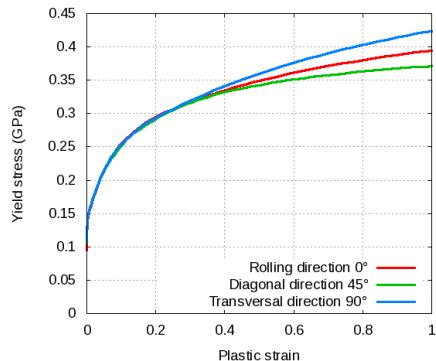
$$\Delta D_{45} = n_{45} D_{45}^{\left(1 - \frac{1}{n_{45}}\right)} \frac{\Delta\varepsilon_{45}^{eq}}{\varepsilon_{45}^f(\eta)}$$



# Example

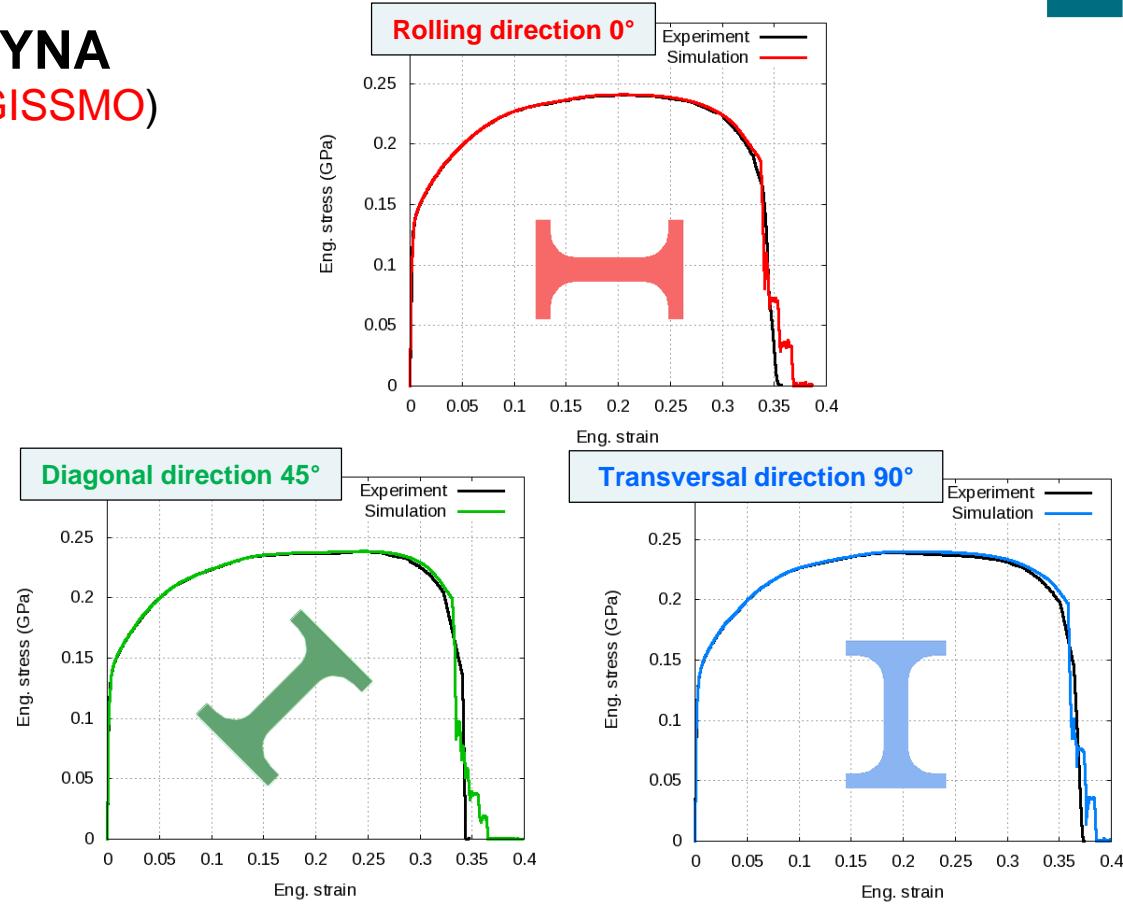
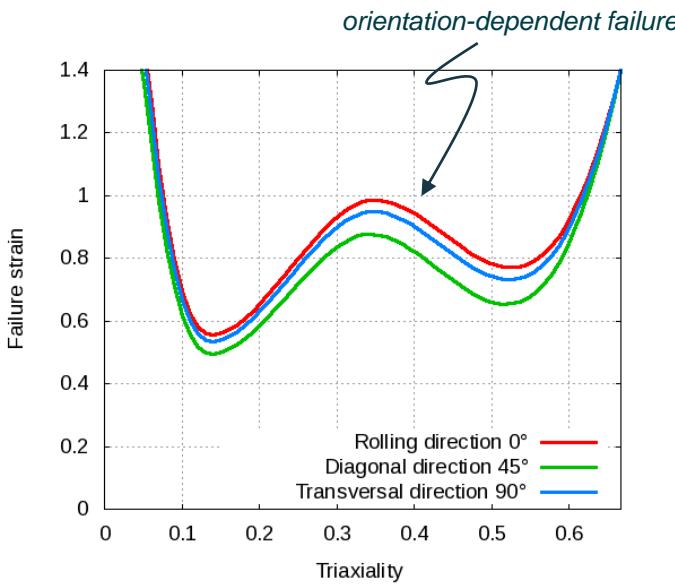
# Material modeling in LS-DYNA

Aluminum sheet (\*MAT\_036E, no failure)



# Material modeling in LS-DYNA

Aluminum sheet (\*MAT\_036E + eGISSMO)



# Final remarks

- A reasonable description of orthotropic plasticity is crucial for accurate plastic strains and, therefore, for an accurate failure prediction.
- In case of orthotropic damage, special components of the plastic strain tensor are evaluated. These are the drivers for the damage accumulation in three material directions.
- \*MAT\_ADD\_GENERALIZED\_DAMAGE / eGISSMO is a highly flexible damage/failure model that can consider orthotropic damage, among other things.
- eGISSMO may also consider damage due to other contributions (e.g., deviatoric and volumetric splitting). Consider a mighty constitutive toolbox!
- eGISSMO is available since **LS-DYNA version R9**.



# eGISSMO

Give it a chance!  
We will support you.